

Thin film growth and properties of nanostructured thermal barrier coatings

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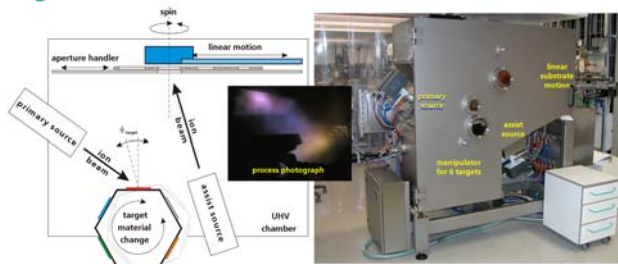
Introduction

Typical demands on thermal barrier coatings (TBC's) are low thermal conductivity and exceptional thermal stability at the operating temperatures they are designed for. Nanostructured materials can improve the thermal as well as the mechanical properties of a TBC in comparison to standard single layer coatings. We show two applications of tailored thermal barriers introducing a laminated micro- and nanostructure.

- A combined Pulsed-Laser-Deposition (PLD) technique is used to prepare columnar-lamellar ZrO_2 coatings that can sustain thermal heat flows as high as 100 MW/m^2 typical for rocket jet engines and that are specially adapted to the loading profile in the engine.
- Nanolaminates of W/Al_2O_3 were produced by Ion Beam Sputter Deposition (IBSD). This laminates with individual layers of only a few nanometers and total thicknesses of up to $2,5 \mu\text{m}$ show a low thermal conductivity and a excellent mechanical stability and adhesion. Therefore coatings on flexible metal foils could be realized.

Fabrication of nanostructured TBC's

Large-area dual IBSD



Schematic view of the arrangement for the large-area IBSD

Photograph of the IBSD machine

Technical data:

Ion beam sources:

- primary for sputtering
- secondary for assisting and etching
- excitation principle: ECR = electron cyclotron resonance
- grid size: $400 \times 200 \text{ mm}^2$
- ion energies: $E = 50 - 2000 \text{ eV}$

Vacuum:

- process chamber: $p < 2 \cdot 10^{-8} \text{ mbar}$
- load lock: $p < 5 \cdot 10^{-7} \text{ mbar}$

Targets:

- number: 6 pieces
- size: $400 \times 200 \text{ mm}^2$

Substrates:

- round, up to $\varnothing = 200 \text{ mm}$
- rectangular, up to $L = 500 \text{ mm}$ (without spin)

Apertures:

- for beam shaping and film thickness homogenization
- 4 pieces, automatically changeable

Internal coating by hybrid PLD

Technical data:

- Lasers: ablation: q-switched Nd:YAG; $\lambda = 1064/355 \text{ nm}$;

evaporation: $f = 10 \text{ Hz}$; $E_{\text{pulse}} = 50..1200 \text{ mJ}$; $\tau = 8 \text{ ns}$
 1 kW cw-Nd:YAG ; $\lambda = 1064 \text{ nm}$;
 $f = 1..1000 \text{ Hz}$; $E_{\text{pulse}} = 0,5..4 \text{ J}$;
 $\tau = 0,5..10 \text{ ms}$

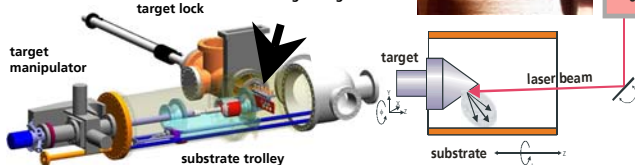
- Vacuum: UHV- $p \approx 10^{-6} \text{ mbar}$

- Targets: cone shaped or foils

- Substrates: $\varnothing_i = 7..120 \text{ mm}$; cylindrical or shaped

$l_{\text{max}} = 60 \text{ cm}$

target magazine

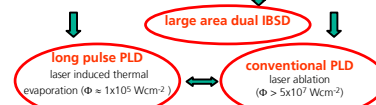
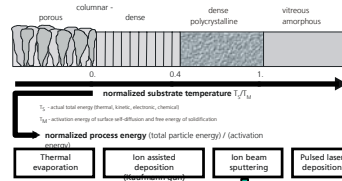


UHV-chamber for internal coating by PLD

Scheme of the internal PLD process

Structure zone model of thin film growth

Extended structure zone model of thin film growth:
(Based on K.H. Guenther, SPIE Vol.1324, 1990)

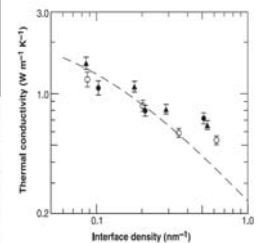
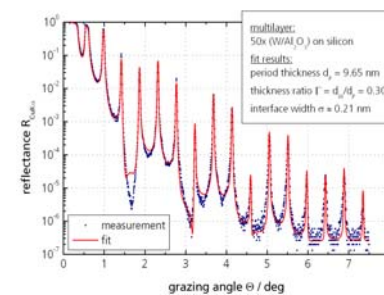


tailored film properties:

- wide range of layer thickness possible: $d = 1 \text{ nm} \dots 100 \mu\text{m}$
- microstructure can be influenced by process energy (e.g. laser parameters, ion energies)
=> amorphous as well as columnar-porous films can be deposited
- nanometer layered structures with ultra high precision
- wide material choice (e.g. metal/non-metal combinations)

Applications

W/Al_2O_3 Nanolaminates



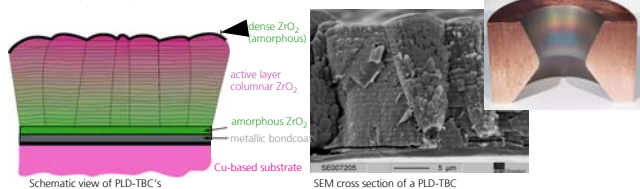
Cu-K α reflectometry of a typical W/Al_2O_3 multilayer with high regularity and a period thickness of $d_p = 9.65 \text{ nm}$ (interface density $1/\delta = 0.21 \text{ nm}^{-1}$). The red curve shows the best fit calculation with an interface width $\sigma \approx 0.21 \text{ nm}$. These sharp transitions between metal and non-metal material act as additional thermal barriers and therefore reduce the thermal conductivity of the nanolaminate.

Room temperature thermal conductivity of W/Al_2O_3 nanolaminates as a function of interface density $1/\delta$; δ is the distance between interfaces. Different symbols represent various deposition techniques (from [1]).

[1]: R.M. Costescu et al.: "Ultra-low thermal conductivity in W/Al_2O_3 nanolaminates", Science 303 (2004) 989

ZrO_2 -coatings for rocket combustion chambers

Coating design:



Experimental results:

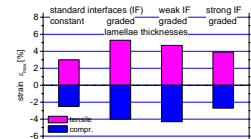
Hot Gas Tests

- tests in model thrust chambers at near working conditions

- LH_2 cooled subscale runs show effective heat load reduction and excellent adhesion of the TBC



Adhesion studies by bending tests



- improved strain resistance with graded and tailored interfaces

